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## **Mechanical analysis of eccentric defected bilayer graphene sheets considering the van der Waals force**

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### **ABSTRACT**

In this paper, it is tried to simulate nonlinear bending analysis of a double-layered graphene sheet (DLGSs) which contains a geometrical imperfection based on an eccentric hole. The first-order shear deformation theory is considered to obtain the governing equations. Also, the nonlinear von Kármán strain field has been assumed in order to obtain large deformations. Whereas the DLGS has been considered, the effect of van der Waals forces has been taken into account in the analysis. In order to implement the nanoscale impact, the nonlocal elasticity theory has been employed. The solution methodology which is here based on the semi-analytical polynomial method (SAPM) solving technique presented previously by the authors, has been applied and again its efficiency has been demonstrated due to its highly accurate results. Due to the fact that this

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research has been done for the first time and there is not any validation available, the results of the local single layer sheet are compared with ABAQUS software. The effects of some other parameters on the results have been studied such as the value of eccentricity, van der Waals interaction and nonlocal parameter.

**Keywords:** Eccentric vacant defect; Nonlocal elasticity theory; First-order shear deformation theory; Bilayer graphene sheet; Van der Waals interaction

### Nomenclature

$w_1$ : Deflection of upper layer	$e$ : Distance between the center of plate and defect
$w_2$ : Deflection of bottom layer	$u$ : In-plane displacement of the plates' nodes along $r$
$r_o$ : Outer radius of the plate	$v$ : In-plane displacement of the plates' nodes along $\theta$
$r_i$ : Size of defect	$w$ : Transverse displacement of the plates' nodes
$\nu$ : Poisson's ratio	$\phi$ : Rotation of elements around circumferential axis
$E$ : Young's modulus	$\psi$ : Rotation of elements around radial axis
$h$ : Thickness of the each layer	$M_r$ : Moment stress resultant around $r$ axis
$k_0$ : van der Waals force	$M_\theta$ : Circumferential moment stress resultant
$q$ : Static uniform transverse load	$M_{r\theta}$ : In-plane moment stress resultant
$k_p$ : Shear layer of elastic matrix	$N_\theta$ : Circumferential in-plane stress resultant
$k_w$ : Winkler modulus of elastic matrix	$N_{r\theta}$ : Torsional in-plane stress resultant
$t$ : time	$N_r$ : Radial in-plane stress resultant
$a$ : an unknown variable in SAPM	$Q_r$ and $Q_\theta$ : Shear stress resultants in two planes
$\mu=(e_0a)^2$ : Nonlocal parameter term	$R_v$ : Ratio of deflection in the presence of defect to the ignoring of defect
$R_m$ : The nonlocal to local analysis	Index $i$ : 1 and 2 for upper and lower layers
$\varepsilon_r$ : Radial strain	$\gamma_{rz}$ : Shear strain in r-z plane
$\varepsilon_\theta$ : Circumferential strain	$\gamma_{r\theta}$ : Shear strain in r- $\theta$ plane



$z$ : Thickness coordinate

$\gamma_{\theta z}$ : Shear strain in  $\theta$ - $z$  plan

$\kappa_s$ : Shear correction factor

$\nabla^2$ : Laplace operator

## 1. Introduction

Double-layered graphene sheets (DLGSs) can be formed from two separated graphene layers which are connected due to the van der Waals interactions between the layers. It is possible to make a bilayer graphene sheet by applying chemical vapor deposition. Many methods have been used to predict the mechanical behavior of nanoscale structures. There are different three methods in this case: atomistic methods, nonlocal continuum methods, and atomistic-continuum methods. If classical mechanics is used, it might lead to considerable errors during the analysis. Between the mentioned theories, nonlocal continuum theory includes several methods as nonlocal elasticity theory [1-3], strain gradient and couple stress theories [4], stress-driven elasticity theory [5, 6], and mixing of the mentioned theories [7, 8], within which the nonlocal elasticity theory is more well-known. Therefore, many researchers have applied this theory to model the mechanical behavior of nanostructures.

In [9, 10] the nonlocal elasticity theory has been presented by Eringen which can be a famous theory in elasticity.

During the last years, many researches have been done to aim to analyze the mechanical behavior of graphene sheets, for example, vibration, buckling and bending analyses. Shen et al. [11] investigated the vibration of single-layered graphene sheet-based nano-mechanical sensor via the nonlocal Kirchhoff plate theory. Zenkour et al. [12] studied thermal analysis of a single-layered graphene sheet (SLGS) embedded in viscoelastic medium. Jiang et al. [13] studied for vibration analysis of single-layered graphene sheet-based mass sensor using Galerkin strip distributed transfer function method.

Dastjerdi et al. [14, 15] investigated the static behavior of monolayer annular/circular and double-layered rectangular nano graphene plate. They received this conclusion that the maximum deflection declines along with the increasing small-scale effects. Dastjerdi and Jabbarzadeh [16] found an approximate single layer equivalent for multi-layer graphene sheets based on nonlocal elasticity theory considering third-order shear deformation theory of the plates (TSDT). Liu et al. [17] discussed the effects of defects on the chemical, electronic, magnetic, and mechanical properties of graphene and addressed the associated challenges and prospects on the future study of defects in graphene and other nanocarbon materials. Dastjerdi et al. [18] derived the constitutive equations of nanoplates embedded in elastic matrix based on Eringen nonlocal elasticity theory and applied both first-order shear deformation theory (FSDT) and higher-order shear deformation theory (HSDT) theory. Moreover, Dastjerdi et al. [19] investigated the eccentric defected monolayer graphene sheets embedded in an elastic matrix. They found that the types of boundary conditions and small-scale effects can affect the results. And also, the value of eccentricity can be added too. Malikan et al. [20] considered bilayer graphene sheets bridged on the elastic medium of Pasternak and analyzed the shear and thermal stability resistance of the FSDT modeled system. Ma et al. [21] studied the effects of magnetic field on thin composite films. They considered bilayer thin films in their analysis and used the Heisenberg model to describe the bilayer film structure. Naderi and Saidi [22] developed the nonlocal constitutive equations of nanostructures. They showed that nonlocal theories could not be used in the presence of surface effects. Mohammadimehr et al. [23] investigated the effects of surface stress on the bending and vibration analyses of SLGSs sheets embedded on an elastic foundation using the energy method. Karimi and Shahidi [24] studied thermo-mechanical bending, vibration and buckling analysis of orthotropic graphene sheets using refined nonlocal elasticity theory. They also considered the effects of



surface energy in their analysis. Ansari et al. [25] examined the stability of SLGSs with different boundary conditions using nonlocal elasticity theory. They evaluated the axial buckling analysis in their study. Lu et al. [26] analyzed the vibration and natural frequencies of a multi-layered graphene sheet based on the classical plate theory and analytical solutions. They investigated the influences of various layerwise tension forces. In a special research, Nazemnezhad et al. [27] modeled the multi-layered graphene sheets (MLGSs) as a sandwich plate via molecular dynamic simulation (MD) and estimated the interlayer shear effects. They confirmed that modelling a MLGS with sandwich plate can result in a highly accurate prediction. Wu and Li [28] formulated a multiple time scale method to analyze the SLGSs embedded in the Pasternak foundation on the basis of Eringen nonlocal elasticity model. In a crucial study, Allahyari and Asgari [29] developed a nonlocal higher-order shear deformation theory to consider vibrations of a DLGS placed in an elastic matrix and thermal environment and also studied the surface effects. Hashemi et al. [30] combined the internal viscosity to the elastic model of DLGSs and evaluated the natural frequencies while the DLGS was rested in a Pasternak medium. The classical plate hypothesis was in conjunction with linear Lagrangian strains as well as nonlocal elasticity theory of Eringen to present the desired mathematical relations. In an applicable research, Ansari et al. [31] conducted use of MD to study some vacancy defects in a rectangular SLGS. More importantly, they investigated fracture in the model. Wang et al. [32] based on the analytical results, discussed the effects of thermal on natural frequencies of a DLGS on the basis of nonlocal elasticity model. In a particular work, Allahyari and Asgari [33] presented a nonlinear wave propagation analysis on the SLGS with assuming circular geometry. Moreover, the effects of environments such as magnetic and temperature were also taken into consideration. They utilized the nonlocal elasticity as well as the modified couple stress theory to justify the small scale influences. Giannopoulos and Avntoulla



[34] analyzed crack and crack features on the tensile strength of graphene in thermal environment by means of MD. In another valuable research, Giannopoulos [35] estimated a circular graphene sheet involving a crack under a vibrational situation using a structural mechanics method in a three-dimensional spring-based domain on the basis of the atomistic models.

The irregularity in the atomic scale structure can occur in the two-dimensional (2D) plane graphene sheet during the manufacturing and production process. We can assume that the graphene sheet is also defective in its atomic scale. Such the irregularity in graphene sheets can be classified in some cases as a heterogeneous defect, topological defects (non-hexagonal shape of arranging atoms) and bonding defects (atoms dislocations, and vacancies) [36-38]. In this paper, the term defect is referred to as the vacancy and missing carbon atoms as a material deviation. In graphene, when one or more carbon atoms are missed, there cannot be a bond between surrounding atoms and some bonds remain in a dangling manner [39] (Fig. 1b).

Studying of defected monolayer circular graphene sheet has been done before by the authors. However, it is too hard to find monolayers in reality. So, in this paper, the bilayer sheet is considered and nonlocal elasticity theory has been applied to obtain the governing equations including the van der Waals interaction between the layers. The position of defect is moving and it is not placed at the center so the axisymmetric problem has been obtained. The obtained equations are solved by applying SAPM for which the efficiency is proved again. Whereas there are not any works available in this case, the results are obtained for different locations of defect, and van der Waals interaction. The results are compared with ABAQUS software as well and good agreements have been achieved after this calculation. Finally, the effects of eccentricity, van der Waals interaction and small-scale influences on the results have been studied.

## **2. Constitutive Equations and solution**

Analysis of single layer defected graphene sheet was studied earlier [19]. In Fig. 1. a double-layered graphene plate is considered which contains an internal eccentric hole. The bilayer sheet is embedded in Winkler-Pasternak elastic medium. The values of Winkler and Pasternak stiffness are  $k_w$  and  $k_p$  respectively. In addition,  $k_o$  refers to the amount of van der Waals interaction between the layers.

Stacking and layering the graphene as 2D sheets of graphite with chemical functionalization make bilayer and multilayer graphene sheets. There is a weak connection between layers when we compared the bonding with the connections of atoms in the plane. One can distinguish this weak bonding can be the van der Waals forces. As a rule, the induced electric polarity makes such bonding. There is no thermal or electrical conductivity along the bonding [40]. In an atomic analysis, the van der Waals forces are attained with respect to a derivative of Lennard-Jones pair potential [41]. One can find that this force is a nonlinear function of interatomic distance. However, the nonlinearity in the behavior of van der Waals forces can be important while we analyze in-phase or out-phase for bilayer graphene sheet and as far as we assumed both layers are rigid together, it seems not important to consider nonlinearly the van der Waals forces. As the van der Waals force is based on the Taylor expansion, we only used the first term of this expansion leads to a linear expression. When using Taylor expansion for the van der Waals force and integrating it over the 2D sheet of graphene, the van der Waals bonding force between upper and lower sheets could be achieved [41, 42]. The distance between two layers is  $0.331 \text{ nm} \leq \text{distance} \leq 0.35 \text{ nm}$  depending on how the layers are stacking on each other while both the layers are normally in an equilibrium condition [41-43].

To access the value of the van der Waals coefficient, the distance between layers plays a major role and the amount of this force directly related to the distance [44]. As mentioned by [43,



44], the value of van der Waals force can be varied numerically in unit GPa/nm for linear expressions and GPa/nm<sup>3</sup> for nonlinear ones. Note that the positive sign for van der Waals force represents the attraction of atoms and the negative sign depicts repulsion. As the scope of this paper is only considering the vacancy imperfection into the DLGSs, hence, we ignore the exact values of the van der Waals coefficient and an arbitrary value is in use whilst it results in rigidity behavior of both layers.

In this study, the van der Waals interaction is simulated by linear spring as  $k_o(w_2 - w_1)$ .

With regard to the Fig. 1a and b, geometric parameters of the defect are  $S$ , and  $e$ , and the relation between the distance of the center of eccentric defect from outer boundary is

$$S = e \left( \sqrt{\left(\frac{r_o}{e}\right)^2 - \sin^2(\theta)} + \cos(\theta) \right).$$

The mathematical expressions of strain-displacement by means of FSDT can be appeared in the nonlinear view as

$$\varepsilon_r = \frac{\partial u}{\partial r} + z \frac{\partial \phi}{\partial r} + \frac{1}{2} \left( \frac{\partial w}{\partial r} \right)^2 \quad (1)$$

$$\varepsilon_\theta = \frac{1}{r} \left( \frac{\partial v}{\partial \theta} + u \right) + \frac{z}{r} \left( \frac{\partial \psi}{\partial \theta} + \phi \right) + \frac{1}{2} \left( \frac{1}{r} \frac{\partial w}{\partial \theta} \right)^2 \quad (2)$$

$$\gamma_{rz} = \frac{\partial w}{\partial r} + \phi \quad (3)$$

$$\gamma_{\theta z} = \frac{1}{r} \frac{\partial w}{\partial \theta} + \psi \quad (4)$$

$$\gamma_{r\theta} = \frac{1}{r} \left( \frac{\partial u}{\partial \theta} + z \frac{\partial \phi}{\partial \theta} - v - z\psi \right) + \frac{\partial v}{\partial r} + z \frac{\partial \psi}{\partial r} + \frac{1}{r} \frac{\partial w}{\partial r} \frac{\partial w}{\partial \theta} \quad (5)$$

The governing equations for the bilayer graphene sheet on the basis of Eqs. (1)-(5), and Hamiltonian, can be presented by neglecting some inconsiderable terms as follows [18] ( $i=1, 2$ ):



$$\delta u_i : Ni_{r,r} + \frac{1}{r} (Ni_{r\theta,\theta} + Ni_r - Ni_\theta) = 0 \quad (6)$$

$$\delta v_i : Ni_{r\theta,r} + \frac{1}{r} (Ni_{\theta,\theta} + 2Ni_{r\theta}) = 0 \quad (7)$$

$$\begin{aligned} \delta w_1 : Q1_{r,r} + \frac{1}{r} (Q1_{\theta,\theta} + Q1_r) + q + k_o (w_2 - w_1) + N1_r \frac{\partial^2 w_1}{\partial r^2} \\ + N1_\theta \left( \frac{1}{r} \frac{\partial w_1}{\partial r} + \frac{1}{r^2} \frac{\partial^2 w_1}{\partial \theta^2} \right) + 2N1_{r\theta} \left( \frac{1}{r} \frac{\partial^2 w_1}{\partial r \partial \theta} - \frac{1}{r^2} \frac{\partial w_1}{\partial \theta} \right) = 0 \end{aligned} \quad (8)$$

$$\begin{aligned} \delta w_2 : Q2_{r,r} + \frac{1}{r} (Q2_{\theta,\theta} + Q2_r) - k_o (w_2 - w_1) - k_w w_2 + k_p \nabla^2 w_2 + \\ N2_r \frac{\partial^2 w_2}{\partial r^2} + N2_\theta \left( \frac{1}{r} \frac{\partial w_2}{\partial r} + \frac{1}{r^2} \frac{\partial^2 w_2}{\partial \theta^2} \right) + 2N2_{r\theta} \left( \frac{1}{r} \frac{\partial^2 w_2}{\partial r \partial \theta} - \frac{1}{r^2} \frac{\partial w_2}{\partial \theta} \right) = 0 \end{aligned} \quad (9)$$

$$\delta \phi_i : Mi_{r,r} + \frac{1}{r} (Mi_{r\theta,\theta} + Mi_r - Mi_\theta) - Qi_r = 0 \quad (10)$$

$$\delta \psi_i : Mi_{r\theta,r} + \frac{1}{r} (Mi_{\theta,\theta} + 2Mi_{r\theta}) - Qi_\theta = 0 \quad (11)$$

in which

$$Ni_r = \frac{Eh}{1-\nu^2} \left( \frac{\partial u_i}{\partial r} + \frac{1}{2} \left( \frac{\partial w_i}{\partial r} \right)^2 + \nu \left( \frac{u_i}{r} + \frac{1}{r} \frac{\partial v_i}{\partial \theta} + \frac{1}{2} \left( \frac{1}{r} \frac{\partial w_i}{\partial \theta} \right)^2 \right) \right) \quad (12)$$

$$Ni_\theta = \frac{Eh}{1-\nu^2} \left( \nu \left( \frac{\partial u_i}{\partial r} + \frac{1}{2} \left( \frac{\partial w_i}{\partial r} \right)^2 \right) + \frac{u_i}{r} + \frac{1}{r} \frac{\partial v_i}{\partial \theta} + \frac{1}{2} \left( \frac{1}{r} \frac{\partial w_i}{\partial \theta} \right)^2 \right) \quad (13)$$

$$Ni_{r\theta} = \frac{Eh}{2(1+\nu)} \left( \frac{\partial v_i}{\partial r} - \frac{v_i}{r} + \frac{1}{r} \frac{\partial u_i}{\partial \theta} + \frac{1}{r} \frac{\partial w_i}{\partial r} \frac{\partial w_i}{\partial \theta} \right) \quad (14)$$

$$Qi_r = \frac{\kappa_s Eh}{2(1+\nu)} \left( \frac{\partial w_i}{\partial r} + \phi_i \right) \quad (15)$$

$$Qi_\theta = \frac{\kappa_s Eh}{2(1+\nu)} \left( \frac{1}{r} \frac{\partial w_i}{\partial \theta} + \psi_i \right) \quad (16)$$

$$Mi_r = \frac{Eh^3}{12(1-\nu^2)} \left( \frac{\partial \phi_i}{\partial r} + \frac{\nu}{r} \left( \frac{\partial \psi_i}{\partial \theta} + \phi_i \right) \right) \quad (17)$$

$$Mi_\theta = \frac{Eh^3}{12(1-\nu^2)} \left( \nu \frac{\partial \phi_i}{\partial r} + \frac{1}{r} \left( \frac{\partial \psi_i}{\partial \theta} + \phi_i \right) \right) \quad (18)$$

$$M_{r\theta} = \frac{Eh^3}{24(1+\nu)} \left( \frac{\partial \psi_i}{\partial r} + \frac{1}{r} \frac{\partial \phi_i}{\partial \theta} - \frac{\psi_i}{r} \right) \quad (19)$$

As this paper is concerned with Eringen's nonlocal elasticity theory to implement the small scale effects, the nonlocal stress can be appeared as [14, 45-48]

$$(1 - \mu \nabla^2) \sigma_{ij} = C_{ijkl} \varepsilon_{kl} \quad (20)$$

Therefore, applying Eq. (20) on Eqs. (12)-(19) and re-writing Eqs. (6)-(11), one can get

$$Ni_{r,r} + \frac{1}{r} (Ni_{r\theta,\theta} + Ni_r - Ni_\theta) = 0 \quad (21)$$

$$Ni_{r\theta,r} + \frac{1}{r} (Ni_{\theta,\theta} + 2Ni_{r\theta}) = 0 \quad (22)$$

$$Q1_{r,r} + \frac{1}{r} (Q1_{\theta,\theta} + Q1_r) + (1 - \mu \nabla^2) \left( q + k_o (w2 - w1) + N1_r \frac{\partial^2 w1}{\partial r^2} + N1_\theta \left( \frac{1}{r} \frac{\partial w1}{\partial r} + \frac{1}{r^2} \frac{\partial^2 w1}{\partial \theta^2} \right) + 2N1_{r\theta} \left( \frac{1}{r} \frac{\partial^2 w1}{\partial r \partial \theta} - \frac{1}{r^2} \frac{\partial w1}{\partial \theta} \right) \right) = 0 \quad (23)$$

$$Q2_{r,r} + \frac{1}{r} (Q2_{\theta,\theta} + Q2_r) + (1 - \mu \nabla^2) \left( -k_o (w2 - w1) - k_w w2 + k_p \nabla^2 w2 + N2_r \frac{\partial^2 w2}{\partial r^2} + N2_\theta \left( \frac{1}{r} \frac{\partial w2}{\partial r} + \frac{1}{r^2} \frac{\partial^2 w2}{\partial \theta^2} \right) + 2N2_{r\theta} \left( \frac{1}{r} \frac{\partial^2 w2}{\partial r \partial \theta} - \frac{1}{r^2} \frac{\partial w2}{\partial \theta} \right) \right) = 0 \quad (24)$$

$$Mi_{r,r} + \frac{1}{r} (Mi_{r\theta,\theta} + Mi_r - Mi_\theta) - Qi_r = 0 \quad (25)$$

$$Mi_{r\theta,r} + \frac{1}{r} (Mi_{\theta,\theta} + 2Mi_{r\theta}) - Qi_\theta = 0 \quad (26)$$

Based on the Eqs. (12)-(19), Eqs. (21)-(26) would be attained into displacements as below (Note

that in order to shorten and summarize, as Eqs. (23) and (24) are too long, they are not transferred into the displacement mode)

$$\begin{aligned}
 & \frac{Eh}{1-\nu^2} \left( \frac{\partial^2 ui}{\partial r^2} + \frac{\partial^2 wi}{\partial r^2} \frac{\partial wi}{\partial r} + \nu \left( \frac{1}{r} \frac{\partial ui}{\partial r} - ui \frac{1}{r^2} + \frac{1}{r} \frac{\partial^2 vi}{\partial r \partial \theta} - \frac{1}{r^2} \frac{\partial vi}{\partial \theta} + \frac{1}{r^2} \frac{\partial wi}{\partial \theta} + \frac{1}{r} \frac{\partial^2 wi}{\partial r \partial \theta} \right) \right) \\
 & + \frac{Eh}{2r(1+\nu)} \left( \frac{\partial^2 vi}{\partial r \partial \theta} - \frac{1}{r} \frac{\partial vi}{\partial \theta} + \frac{1}{r} \frac{\partial^2 ui}{\partial \theta^2} + \frac{1}{r} \frac{\partial wi}{\partial r} \frac{\partial^2 wi}{\partial \theta^2} + \frac{1}{r} \frac{\partial^2 wi}{\partial r \partial \theta} \frac{\partial wi}{\partial \theta} \right) \\
 & + \frac{Eh}{r(1-\nu^2)} \left( \frac{\partial ui}{\partial r} + \frac{1}{2} \left( \frac{\partial wi}{\partial r} \right)^2 + \nu \left( \frac{ui}{r} + \frac{1}{r} \frac{\partial vi}{\partial \theta} + \frac{1}{2} \left( \frac{1}{r} \frac{\partial wi}{\partial \theta} \right)^2 \right) \right) \\
 & - \frac{Eh}{r(1-\nu^2)} \left( \nu \left( \frac{\partial ui}{\partial r} + \frac{1}{2} \left( \frac{\partial wi}{\partial r} \right)^2 \right) + \frac{ui}{r} + \frac{1}{r} \frac{\partial vi}{\partial \theta} + \frac{1}{2} \left( \frac{1}{r} \frac{\partial wi}{\partial \theta} \right)^2 \right) = 0
 \end{aligned} \tag{27}$$

$$\begin{aligned}
 & \frac{Eh}{2(1+\nu)} \left( \frac{\partial^2 vi}{\partial r^2} - \frac{1}{r} \frac{\partial vi}{\partial r} + \frac{vi}{r^2} - \frac{1}{r^2} \frac{\partial ui}{\partial \theta} + \frac{1}{r} \frac{\partial^2 ui}{\partial r \partial \theta} - \frac{1}{r^2} \frac{\partial wi}{\partial r} \frac{\partial wi}{\partial \theta} + \frac{1}{r} \frac{\partial^2 wi}{\partial r \partial \theta} \frac{\partial wi}{\partial \theta} + \frac{1}{r} \frac{\partial^2 wi}{\partial r^2} \frac{\partial wi}{\partial \theta} \right) \\
 & + \frac{Eh}{r(1-\nu^2)} \left( \nu \left( \frac{\partial^2 ui}{\partial r \partial \theta} + \frac{\partial wi}{\partial r} \frac{\partial^2 wi}{\partial r \partial \theta} \right) + \frac{1}{r} \frac{\partial ui}{\partial \theta} + \frac{1}{r} \frac{\partial^2 vi}{\partial \theta^2} + \frac{1}{r} \frac{\partial wi}{\partial \theta} \frac{\partial^2 wi}{\partial \theta^2} \right) \\
 & + \frac{Eh}{r(1+\nu)} \left( \frac{\partial vi}{\partial r} - \frac{vi}{r} + \frac{1}{r} \frac{\partial ui}{\partial \theta} + \frac{1}{r} \frac{\partial wi}{\partial r} \frac{\partial wi}{\partial \theta} \right) = 0
 \end{aligned} \tag{28}$$

$$\begin{aligned}
 & \frac{Eh^3}{12(1-\nu^2)} \left( \frac{\partial^2 \phi i}{\partial r^2} - \frac{\nu}{r^2} \left( \frac{\partial \psi i}{\partial \theta} + \phi i \right) + \frac{\nu}{r} \left( \frac{\partial^2 \psi i}{\partial r \partial \theta} + \frac{\partial \phi i}{\partial r} \right) \right) \\
 & + \frac{Eh^3}{24r(1+\nu)} \left( \frac{\partial^2 \psi i}{\partial r \partial \theta} + \frac{1}{r} \frac{\partial^2 \phi i}{\partial \theta^2} - \frac{1}{r} \frac{\partial \psi i}{\partial \theta} \right) + \frac{Eh^3}{12r(1-\nu^2)} \left( \frac{\partial \phi i}{\partial r} + \frac{\nu}{r} \left( \frac{\partial \psi i}{\partial \theta} + \phi i \right) \right) \\
 & - \frac{Eh^3}{12r(1-\nu^2)} \left( \nu \frac{\partial \phi i}{\partial r} + \frac{1}{r} \left( \frac{\partial \psi i}{\partial \theta} + \phi i \right) \right) - \frac{\kappa_s Eh}{2(1+\nu)} \left( \frac{\partial wi}{\partial r} + \phi i \right) = 0
 \end{aligned} \tag{29}$$

$$\begin{aligned}
 & \frac{Eh^3}{24(1+\nu)} \left( \frac{\partial^2 \psi i}{\partial r^2} - \frac{1}{r^2} \frac{\partial \phi i}{\partial \theta} + \frac{1}{r} \frac{\partial^2 \phi i}{\partial r \partial \theta} + \frac{\psi i}{r^2} - \frac{1}{r} \frac{\partial \psi i}{\partial r} \right) - \frac{\kappa_s Eh}{2(1+\nu)} \left( \frac{1}{r} \frac{\partial wi}{\partial \theta} + \psi i \right) \\
 & + \frac{Eh^3}{12r(1-\nu^2)} \left( \nu \frac{\partial^2 \phi i}{\partial r \partial \theta} + \frac{1}{r} \left( \frac{\partial^2 \psi i}{\partial \theta^2} + \frac{\partial \phi i}{\partial \theta} \right) \right) + \frac{Eh^3}{12r(1+\nu)} \left( \frac{\partial \psi i}{\partial r} + \frac{1}{r} \frac{\partial \phi i}{\partial \theta} - \frac{\psi i}{r} \right) = 0
 \end{aligned} \tag{30}$$

Taking a look at the governing equations, it is clear that a two-dimensional solving process must be considered in which its solution is not possible by common analytical methods. In the paper, due to eccentricity and axisymmetric nonlinear problems, applying common analytical and

numerical methods are not possible, hence SAPM method which is presented by the authors [14, 49] before is applied again. According to the previous work, the polynomial functions can be presented below: ( $i=1, 2$ )

$$ui = \sum_{k=1}^N \sum_{t=1}^M ai_{(k+t-(1-(k-1)(M-1)))} r^{(k-1)} \theta^{(t-1)} \quad (31)$$

$$vi = \sum_{k=1}^N \sum_{t=1}^M ai_{(k+t-(1-(k-1)(M-1))+M \cdot N)} r^{(k-1)} \theta^{(t-1)} \quad (32)$$

$$wi = \sum_{k=1}^N \sum_{t=1}^M ai_{(k+t-(1-(k-1)(M-1))+2M \cdot N)} r^{(k-1)} \theta^{(t-1)} \quad (33)$$

$$\phi i = \sum_{k=1}^N \sum_{t=1}^M ai_{(k+t-(1-(k-1)(M-1))+3M \cdot N)} r^{(k-1)} \theta^{(t-1)} \quad (34)$$

$$\psi i = \sum_{k=1}^N \sum_{t=1}^M ai_{(k+t-(1-(k-1)(M-1))+4M \cdot N)} r^{(k-1)} \theta^{(t-1)} \quad (35)$$

By considering the above polynomial equations, the constitutive equations are transformed into algebraic equations system and might be solved conveniently. The obtained equations can be solved by the definition of boundary conditions.

### 3. Numerical results

Owing to this problem that there is no available study on the effect of eccentric defects, the obtained results of the monolayer macro sheet based on local theory considering different eccentricity values are validated with ABAQUS in Table 1. To assess the problem in FEM, we use a nonlinear triangular element. Moreover, the boundary conditions in the software were taken as CC and CF, which the first character relates to the outer diameter and the second one dedicates to the inner diameter. Generally, in this paper the C boundary condition means clamped and the F means free edges. The SAPM method is approximately 40 percent faster than ABAQUS by

considering high number of elements. Also, the nonlocal results of this paper for an annular/circular graphene sheet is compared with Ref [14] in Table 2. The insignificant differences between the results of this paper and Ref [14] are observed because this paper is two-dimensional analysis and Ref [14] is one dimensional. It is observed that the achieved results are similar to the results of ABAQUS and Ref [14]. Consequently, the SAPM gives acceptable results in this case and it could be possible to validate the obtained results of the mentioned method with the other references. The geometry and material properties of the sheet could be seen below (Table 1):

$$r_i = 0.1\text{m}; r_o = 0.5\text{m}; h = 0.01\text{m}; E = 1.9 \times 10^{11} \text{ N/m}^2; \nu = 0.29; q = 10^4 \text{ Pa}$$

In Fig. 2, deflection of two layers versus the van der Waals interaction between two layers are compared. The plate specifications are as follows [19]:

$$r_i = 1\text{nm}; r_o = 5\text{nm}; h = 0.34\text{nm}; e = 1\text{nm}; E = 1.06\text{TPa}; \nu = 0.3; q = 0.1\text{GPa}$$

$$k_w = 1.13\text{GPa/nm}; k_p = 1.13\text{Pa.m}$$

According to Fig. 2, it is observed that by raising the van der Waals interaction between two layers, the deflection of the upper layer decreases and for the bottom layer increases. In the following, by raising van der Waals force, deflections of both layers reach to each other, in continue, no significant differences can be seen. It can be noted that the van der Waals force between layers is dependent on the distance between layers and usually is taken  $k_o = 40 \text{ GPa/nm}$ . As Fig. 2 is shown, by raising the nonlocal parameter the deflections of both upper and bottom layers reach each other by small values of  $k_o$  coefficient, and it has resulted that by raising nonlocal coefficient, the effect of van der Waals interaction between layers will be increased.

Fig. 3 is plotted to show the effect of  $R_m$ . As it is observed, for the upper layer, deflection in the nonlocal case is always less than the local one. On the other hand, for the upper layer which  $R_m < 1$ , by raising  $k_o$  or raising van der Walls force up to a specified value of  $k_o$ , two nonlocal and

local analyses are far from each other and slope of the curve is steep. However, hereinafter by raising  $k_o$ ,  $R_m$  is constant without any effect, and the slope of the curve is approximately zero. Then, it is observed that by raising nonlocal coefficient, differences of nonlocal and local theories become more than before; however, without any effect on the mentioned  $k_o$  which hereinafter variation of  $R_m$  becomes approximately zero. Also, the results of  $e_0a = 1$  nm and  $e_0a = 2$  nm are similar as well. As it is shown in Fig. 3, for the bottom layer up to a specified  $k_o$  on the contrary of the upper layer which deflection of nonlocal is less than local case, deflection of nonlocal is more than local. The rate of falling  $R_m$  for the bottom layer is similar to the upper layer, but the slope of variations is less than the upper layer. Finally, it is observed that by raising  $k_o$ ,  $R_m$  for both upper and bottom layers is identical, and after a specified  $k_o$ , no change in  $R_m$  occurs, and the specified  $k_o$  for two nonlocal cases of  $e_0a = 1$  nm and  $e_0a = 2$  nm is approximately similar.

Fig. 4 (a) and (b) show the diagram of deflection variations of the plate with CC boundary conditions versus raising eccentricity for different values of elastic medium Winkler-Pasternak. Taking a look at diagrams, it is concluded that the deflection of both upper and bottom layers will be increased by raising eccentricity  $e$ . Also, by raising  $k_w$  and  $k_p$  as it is expected, the deflection will be decreased, but it is considered that by raising  $k_p$  the deflection of the plate is reduced a little more than the case with just Winkler medium. Therefore, in this example, the Pasternak medium has more effect on the decrease of deflection. Slopes of both diagrams of Fig. 4 are approximately similar and it has resulted that the effects of raising eccentricity  $e$  for both Winkler and Pasternak are the same with a slight deviation. According to the diagrams, it is observed that by raising  $k_w$  and  $k_p$ , the rate of variation is decreased. On the other hand, the effect of eccentricity  $e$  is decreased and it is concluded that in high values of  $k_w$  and  $k_p$ , the effect of raising of eccentricity  $e$  on results becomes very smaller than before. Also, by raising  $k_w$  and  $k_p$ , deviation of two upper



and bottom layers is increased, due to the bottom layer embedded on elastic medium and the upper layer is under transverse loading. As well as raising  $k_w$  and  $k_p$ , deflection of the bottom layer rather than the upper layer is reduced more, therefore deviation of deflection for both layers is heightened.

In Fig. 5, the diagram shows variation of deflection versus raising eccentricity  $e$ , considering different nonlocal coefficients (CC boundary conditions). According to Fig. 5, as the same as Fig. 4 (a) and (b), it is observed that by raising eccentricity  $e$ , deflections of both layers are raised. Again, by raising nonlocal coefficient, deflections of both layers get similar to each other, but no considerable changes in the slope of diagrams are observed. Further to the mentioned paragraphs, by raising nonlocal coefficient, deflections of both layers get closer and it has resulted that the nonlocal coefficient effects on van der Waals interaction between layers is increased which leads to strengthening the force between the layers.

Fig. 6 shows the effect of nonlocal analysis versus moving defect. It is observed that by raising eccentricity  $e$ , variations of  $R_m$  are very small. Distinctly, for the upper layer, the changes are very small and ignorable, but for bottom layer the variations are considerable and it is observed that by raising eccentricity  $e$ ,  $R_m$  is decreased and approaches to the value for the upper layer. Therefore, in graphene sheets with two layers, it can be seen that by raising or decreasing the position of defect, no considerable change in the nonlocal analysis is attained. For example, if nonlocal deflection in small values of eccentricity  $e$  is available, the deflection of the plate for higher values of eccentricity ( $e$ ) could be predicted as well. In particular, with increasing the eccentricity of defect, by increasing anti-symmetric specification, the numerical analysis will be more complicated, and according to Fig. 6, the resulted values of small  $e$  could be utilized for obtaining results of the cases with larger  $e$ .

To study the effect of the existence of the defect, Fig. 7 is presented. As it is observed, by

raising the nonlocal coefficient, the value of  $R_v$  is decreased, so increasing the small-scale effects leads to reducing the effect of availability of defect. Another result according to Fig. 7 can be mentioned that the variation of  $R_v$  for the bottom layer is severely more than the upper layer. For the bottom layer, at the start of growing the nonlocal coefficient, the variations are severe, but for the upper layer is vice versa and the variations are small. Also, for the bottom layer, the slope of variations is decreased and by the manner, along the raise of nonlocal coefficient,  $R_v$  is increased approximately similar for both upper and bottom layers.

The effect of defect size in a bilayer defected circular graphene sheet is studied. Fig. 8 is presented in this regard. As it is demonstrated, by increasing defect size, the deflection of the plate is decreased. At first, the variations are severe, but in continue, the variations are decreased. Also, by raising the nonlocal coefficient, as it has resulted before, it is observed that deflection of both upper and bottom layers tends to each other but the slope of variations of two layers are similar and increase of defect size has the same effect on the results of upper and bottom layers.

Finally, the nonlinear analysis has been conducted on the research. The nonlinear terms in the strain field lead to more accurate results for large deflections. Whatever the amount of loading imposed on the sheet surface intensifies, the large deflection will be produced. Consequently, the deflection variation of upper and bottom layers versus loading is attended. Fig. 9 shows the comparison between the linear and nonlinear analyses for a defected bilayer graphene sheet. It is observed that for lower amounts of loadings both linear and nonlinear analyses are identical. However, as the loading increases, the differences between linear and nonlinear studies will be more significant. The linear analysis gives the same increment of the results for the same variation of loading, absolutely on the contrary in nonlinear analysis. The differences between linear and nonlinear analyses are more considerable for the upper layer deflection rather than the bottom





layer. According to Fig. 9, it is concluded that if the sheet is under high loading, the linear analysis causes serious errors and the obtained results are not acceptable and the nonlinear analysis must be applied instead. The nonlinear analysis will be more complicated and increases the number of calculations dramatically; however, as stated before it gives more accurate results. Consequently, in this paper, the nonlinear analysis has been attended.

#### 4. Conclusions

In the present study, the nonlinear bending behavior of double-layered defected graphene sheets is studied based on nonlocal elasticity theory while the plate is rested in an elastic matrix. The governing equations are derived by considering the van der Waals interaction between the layers. The constitutive equations are solved using SAPM which was presented by authors before. The main out-comings of the article can be summarized as follows:

- SAPM which is a semi-analytical method gives appropriate results in comparison with the other numerical methods.
- By raising the nonlocal coefficient, the effect of van der Waals interaction between the layers will be increased.
- The results are dependent on the value of Winkler and Pasternak elastic foundations. Whatever the defect reaches to the edges, the effect of Winkler and Pasternak elastic mediums are approximately the same, however, the Winkler foundation's effect is inconsiderably more than Pasternak.
- The value of eccentricity has the same effects on the results for different values of the nonlocal parameter.
- The increase of defect size has the same effects on the results of the upper and bottom layers.

- The presence of the defect affects the upper layer's deflection significantly more than the bottom layer.
- The small-scale effect decreases with the growth of the defect.
- Nonlinear analysis gives more accurate results especially for large deflections in which the linear results are not acceptable.

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## Figure Captions

**Fig. 1.** Schematic view of Winkler-Pasternak medium and van der Waals interaction (a) and geometrical view of the defect (b)

**Fig. 2.** Maximum deflection of upper and bottom layers versus  $k_o$

**Fig. 3.**  $R_m$  for upper and bottom layers versus  $k_o$

**Fig. 4.** Maximum deflection of upper and bottom layers versus eccentricity for (a) different values of  $k_w$  (b)  $k_p$

**Fig. 5.** Maximum deflection of upper and bottom layers versus eccentricity for different values of nonlocal parameter

**Fig. 6.** Variation of  $R_m$  for upper and bottom layers versus eccentricity for different values of nonlocal parameter

**Fig. 7.** Variation of  $R_v$  for upper and bottom layers versus nonlocal parameter

**Fig. 8.** Variation of maximum deflection for upper and bottom layers versus  $r_i$  for different values of nonlocal parameter

**Fig. 9.** Linear and nonlinear deflection results of bilayer defected graphene sheet

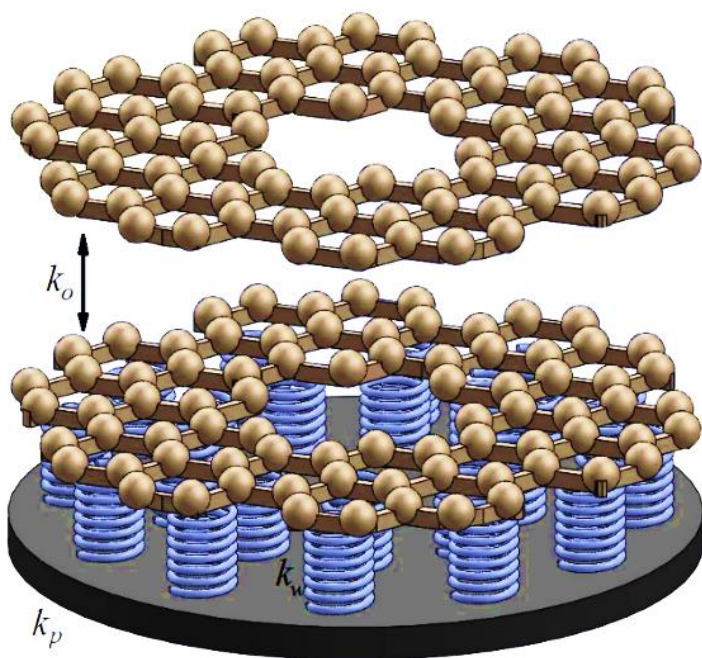


## List of Tables

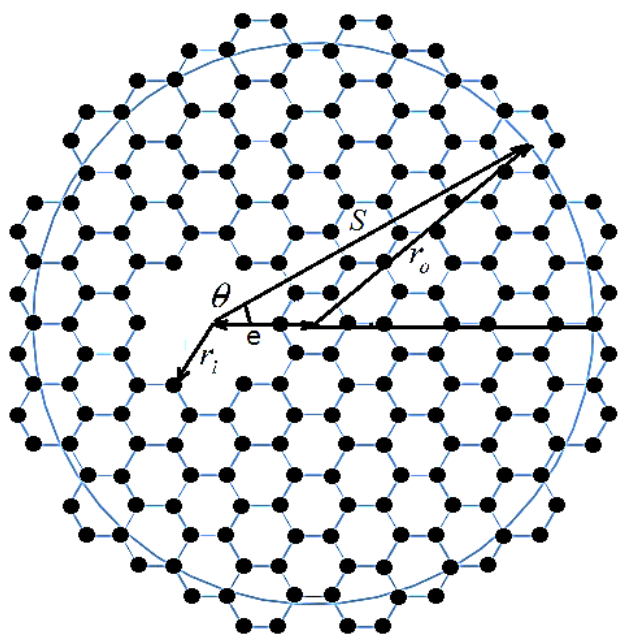
**Table 1.** Maximum deflection obtaining in present paper and ABAQUS

**Table 2.** Comparison between the results of present paper and Ref [14]

Fig. 1

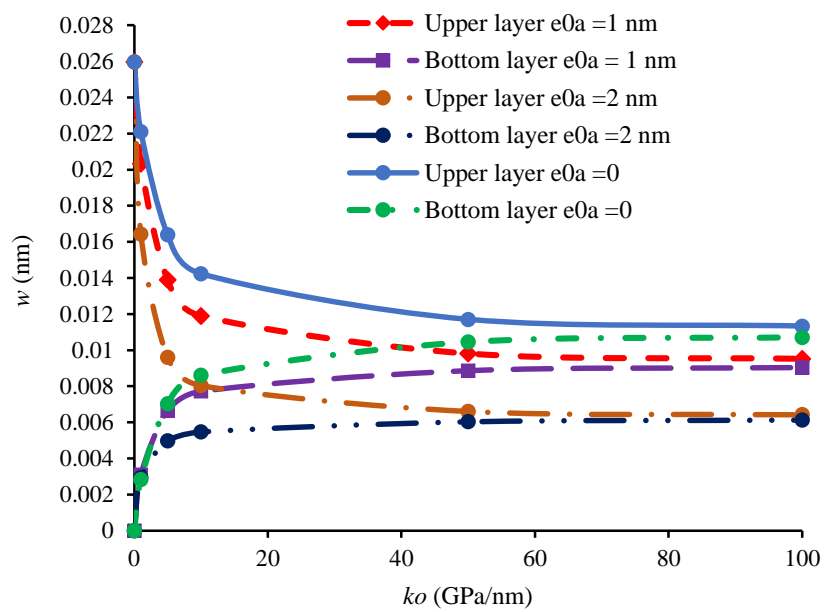


(a)



(b)

**Fig. 2**



**Fig. 3**

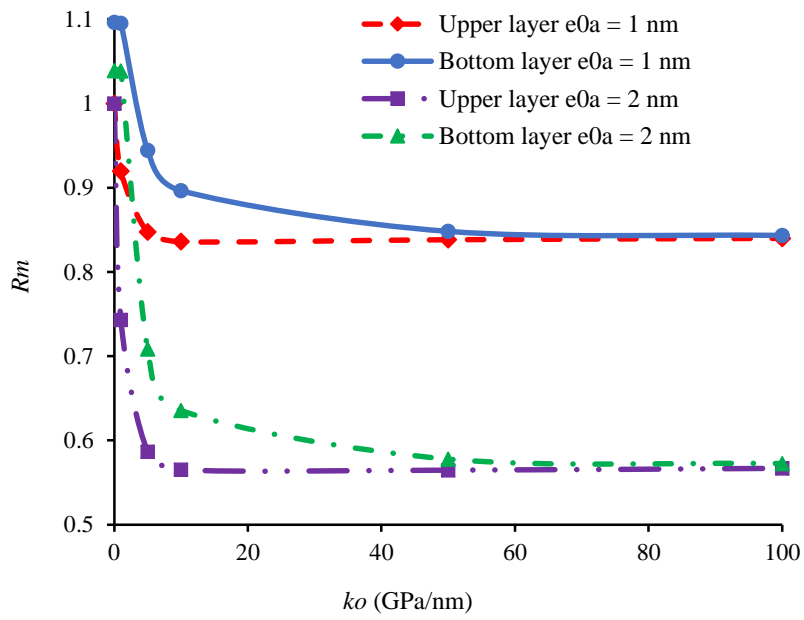
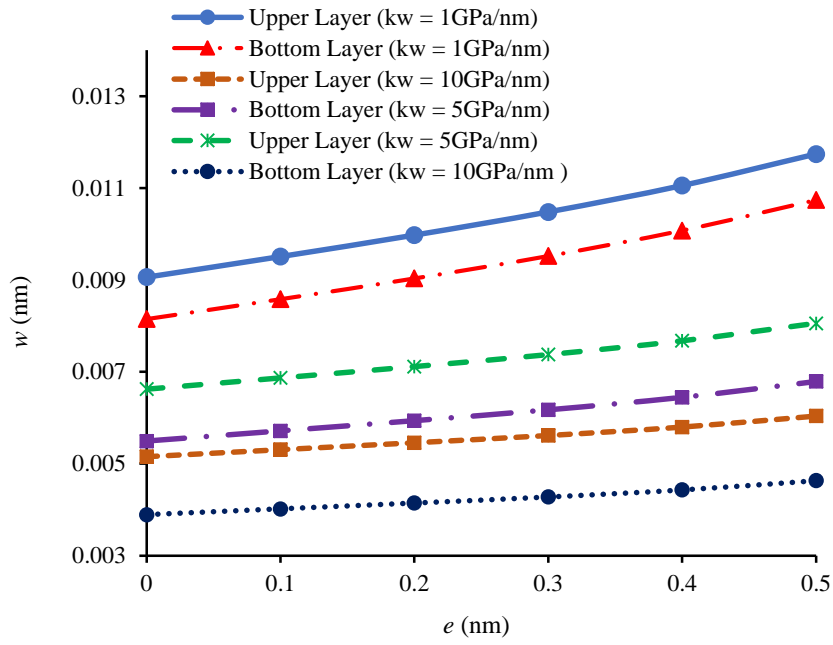
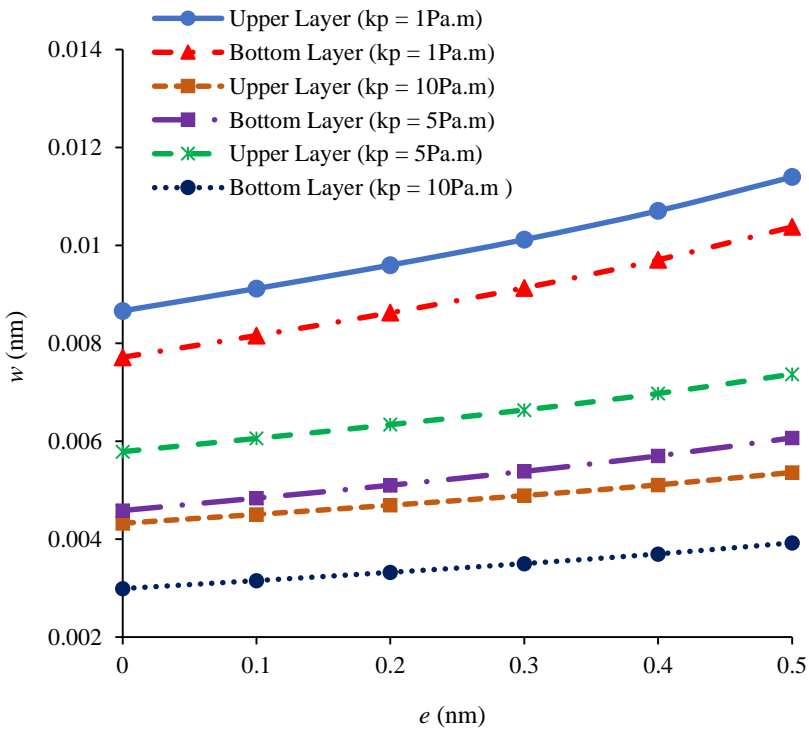


Fig. 4



(a)



(b)

Fig. 5

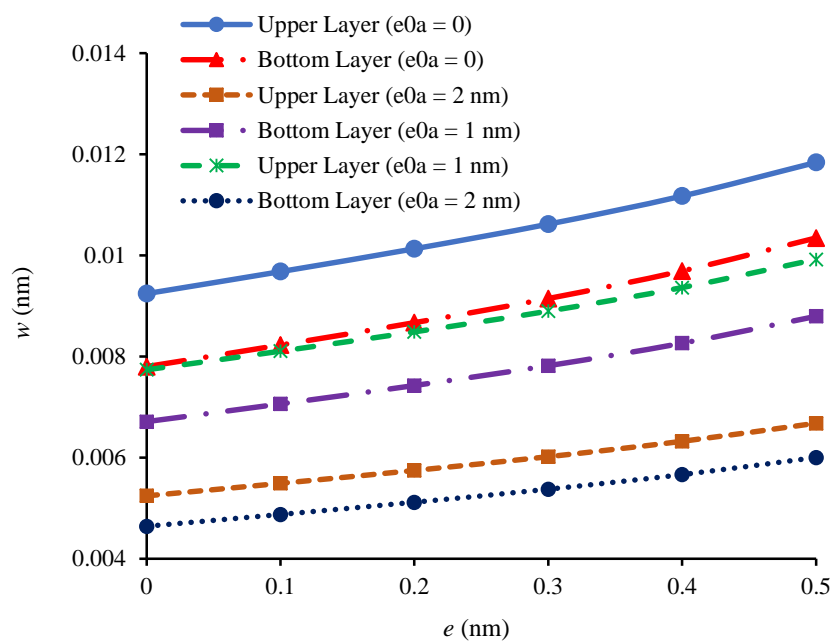
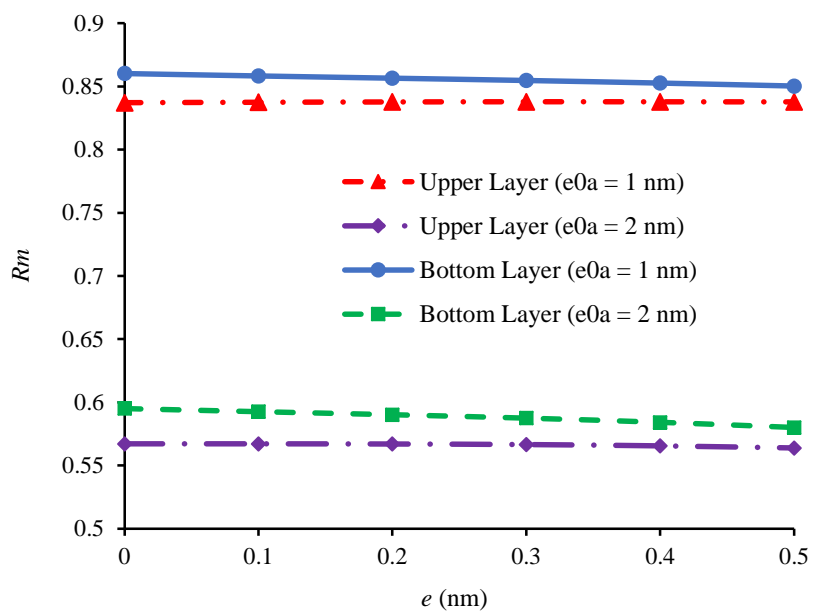
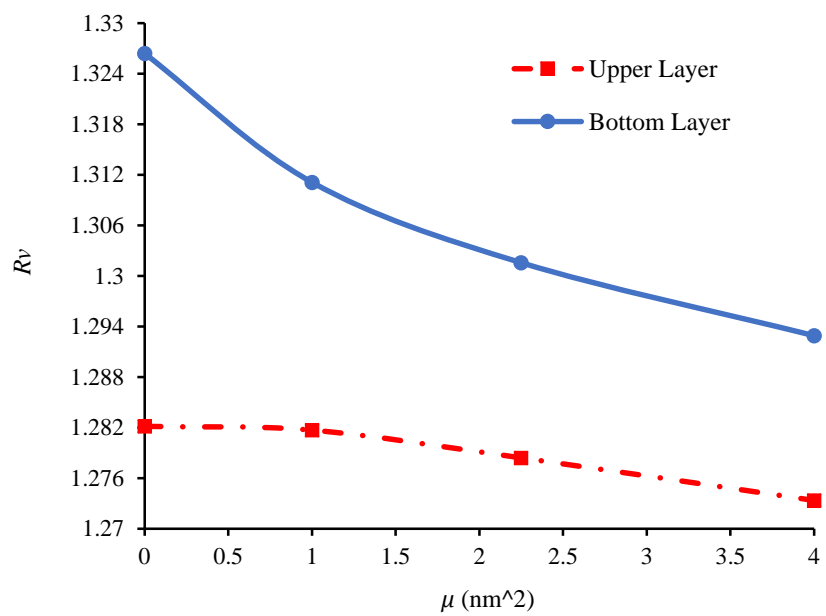


Fig. 6

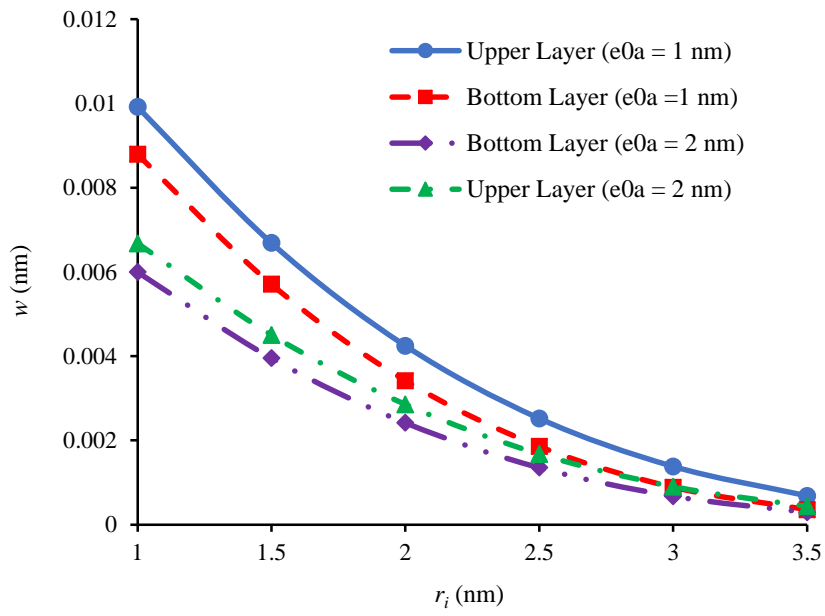


**Fig. 7**

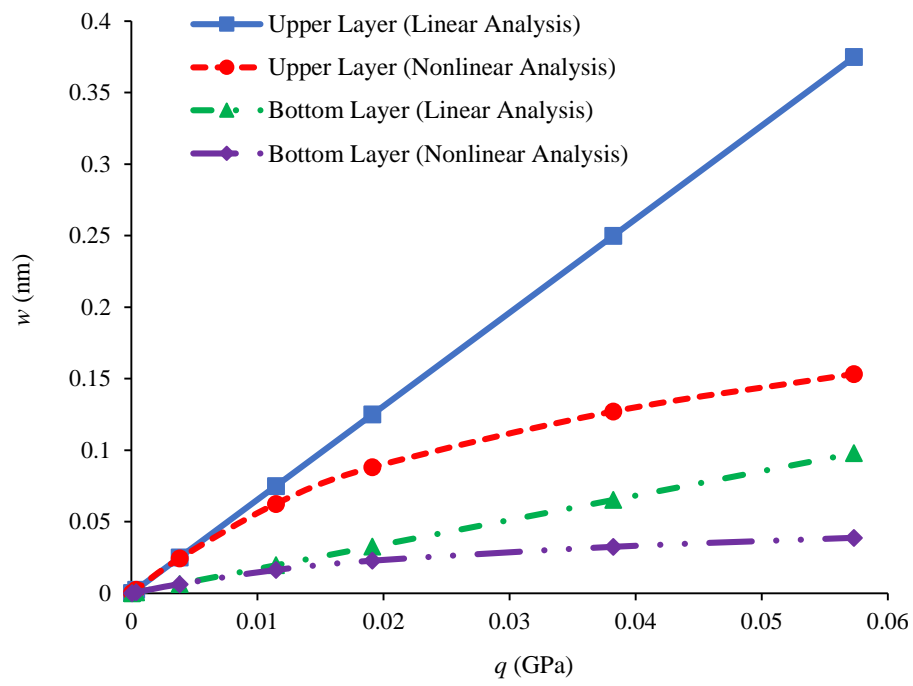




**Fig. 8**



**Fig. 9**



**Table 1.** Maximum deflection obtaining in the present paper and ABAQUS for a circular macro plate

BCs	e (m)	w (mm)				Present
		ABAQUS				
		Elements=6500	13000	30000	50000	
CC	0	0.03862	0.03902	0.03933	0.03933	0.03936
	0.05	0.05972	0.06032	0.06090	0.06090	0.05757
	0.1	0.08747	0.08839	0.08910	0.08910	0.08611
	0.15	0.12250	0.12370	0.12435	0.12435	0.1121
	0.2	0.16460	0.16596	0.16678	0.16678	0.1507
FC	0	0.57170	0.57392	0.57592	0.57592	0.5711
	0.05	0.60771	0.61020	0.61261	0.61261	0.5895
	0.1	0.61642	0.61952	0.62213	0.62213	0.6012
	0.15	0.59870	0.60258	0.60484	0.60484	0.5773
	0.2	0.56261	0.56509	0.56712	0.56712	0.5467

**Table 2.** Comparison between the results of present paper and Ref [14]

$e_0 a$ (nm)	$w / r_0$	
	[14]	Present
0	0.003063	0.003069
0.5	0.002985	0.002990
1	0.002774	0.002778
1.5	0.002484	0.002487
2	0.002170	0.002173